

# DESIGN OF THE ORNITHOPTER

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## ABSTRACT

An ornithopter is a type of aircraft that uses flopping bodies to induce lift and propulsion. These designs are inspired by natural fliers similar as catcalls and insects known for their ease of flight and project in the open sky. The conception of an ornithopter has been around for centuries, with Leonardo da Vinci being one of the first to sketch designs for a flying machine that used flopping bodies. moment, ornithopters are frequently studied as a implicit volition to traditional aircraft, as they've the eventuality to be more effective and maneuverable in certain situations.

**Keyword :** - natural aviators, manoeuvrability, surveillance ....

## 1. INTRODUCTION

Since time old, flying in the sky has been a source of seductiveness to utmost humans. The sight of the catcalls flying freely and majestically in the air fills one with craving and joy. And therefore were inspired the colorful attempts by humanity to achieve flight and take to the skies. Natural fliers do with surprising diversity, inhabiting most kinds of territories on earth, and display remarkable evolutionary features that allow them acclimatize expeditiously to their terrain. catcalls have a weight range varying from around 1g( Bee- Humming raspberry) to around 10- 12 kgs in certain large raptors. The size range in catcalls varies from around 5- 6 cm to a humungous wingspan of 1.8- 2 m set up in Philippines eagle. Insects on the other hand have spans varying from a many hundred microns in the lowest insects to around 25 cms of an Atlas Moth. In these ranges due to the low- Reynolds number flows, the conventional aerodynamics isn't accurate enough to prognosticate the geste of the aircraft. Fixed winged aircrafts typically employ bodies only to induce the lift needed to sustain in the air. The thrust needed to overcome drag is typically supplied by a separate propulsion system. This necessitates the presence of an redundant device which can be avoided if lift and thrust can be combined as in nature. In addition, fixed sect aircrafts have numerous other disadvantages as compared to Ornithopters which will be indicated latterly. Another prominent kind of aircraft is the rotary sect aircraft. These have disadvantages in terms of speed, effectiveness and manoeuvrability. That said, the question might arise as to why Ornithopters haven't attained a lot of fashionability. The simple reason for this is the position of complication involved. also the advantages of an Ornithopter are more applicable at small scales, where, unfortunately, incorporating the flopping sect features is mechanically further *of a challenge*

## 2. LITERATURE REVIEW

Fixed winged aircrafts generally employ bodies only to induce the lift demanded to sustain in the air. The thrust demanded to overcome drag is generally supplied by a separate propulsion system. This necessitates the presence of an spare device which can be avoided if lift and thrust can be combined as in nature. In addition, fixed side aircrafts have numerous other disadvantages as compared to Ornithopters which will be indicated latterly.

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### 2.1 Aerodynamics

Review of introductory Aerodynamics is a branch of dynamics concerned with studying the stir of air, particularly when it interacts with a solid object. Aerodynamics is a subfield of fluid dynamics and gas dynamics, with important proposition shared between them. Aerodynamics is constantly used synonymously with gas dynamics, with the difference being that gas dynamics applies to all feasts.

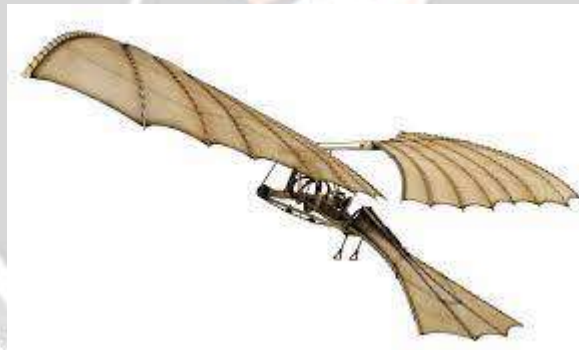
### 2.2 Lift

Lift and the Measure of Lift is the aerodynamic force acting on any body moving through a fluid in a direction perpendicular to the direction of its stir. The product of lift can be explained on the base of Newton " s third law

### 3. PROBLEM DEFINITION AND METHODOLOGY

The introductory methodology used in this model is a flopping side medium. They hold the pivotal features demanded for flight of the prototype. Although it's seen to be impossible to achieve the exact moment of side conduct through mechanical gear of a model certain aspects can be mimicked to insurance working.

Also, indeed when the shape of bodies add on to their behavior in air. For illustration, the shape of a swallow bodies are different from that of a pelican. swallows are quick to cover short distance and sharp turns but pelicans can last longer, this is due to the wingspan. So selection of the side shape is vital pertaining to the type of operation demanded for the model. The selection of material also has a huge effect in working of the prototype. multitudinous factors should be added to induce the most precise type of paraphernalia used.



**Fig.-3.1**



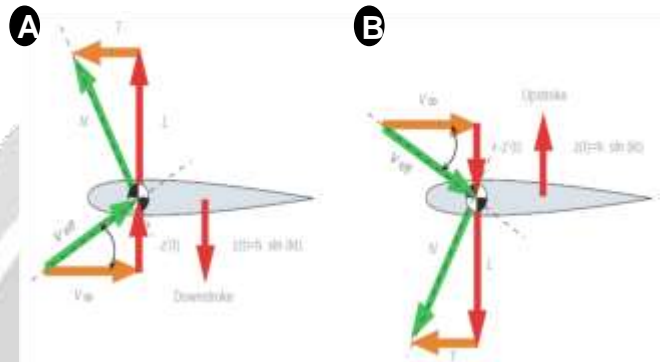
**Fig-3.2**

**Fig-3.1 & 3.2:** are the wing type used in the ornithopter

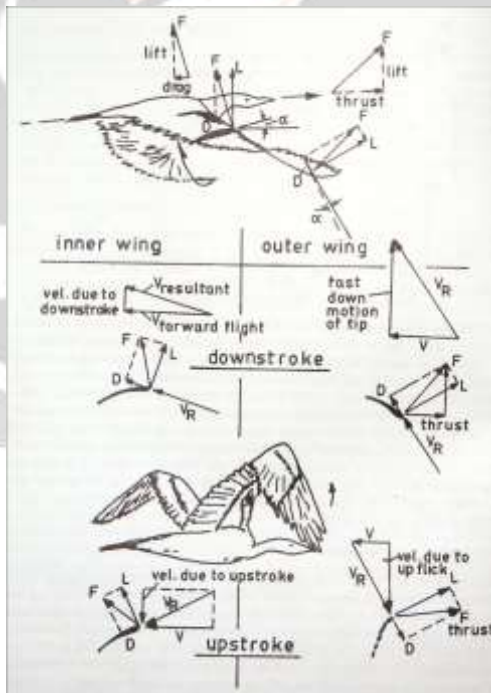
#### 4. Flapping Wing aerodynamics

During flapping flight, the bodies also changes their angle of attack depending on the stroke. flapping flight is principally rowing in the air sect. particularly at the scale of interest is significantly different from fixed bodies substantially.

First at small scale Reynolds number of the model is veritably low compare to macro scale. flapping involves up and down movement of the bodies. Theodorsen assumed that the wake of the air antipode would take the form of a nonstop whirlpool distance of sinusoidally varying strength, stretching from the running edge to perpetuity in the downstream direction. The wake wasn't allowed to change shape in response to the haste convinced by the wake. This Theodorsen's proposition will come in the times to come the standard tool to dissect air antipode flutter, helicopter aerodynamics and flapping flight problems.

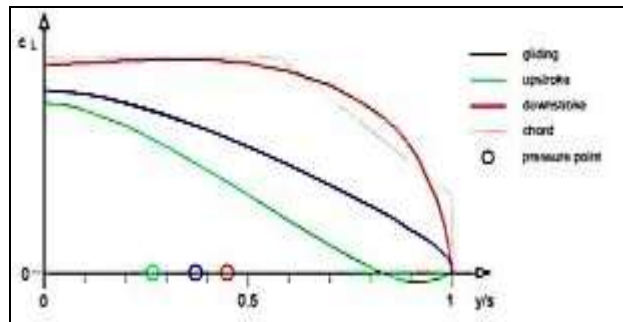


**Fig 4.1:** Thrust (T) and lift (L) components of the normal force vector (N) during heaving motion.



**Fig-4.2:** Wing flapping forces.

**4.1. Distribution curve for flapping wings**

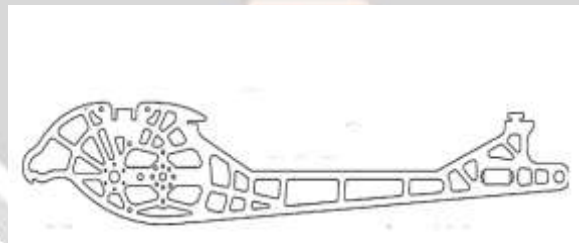


**Fig-4.1:** Force distribution.

These are lift distributions for flopping bodies suitable for “ flying with lift ” in a gently inclined rise flight. In relation to the wingspan the affected a relative le small average necklace around the pivot of the sect. Compare to the electrical distribution then the lift distribution of the gliding flight is more phased. In relation to the bending moment at the sect root and the convinced drag it's optimal for unlimited sect span. At the time elliptical distribution with its 15 lower sect span. The lift distribution of the upstroke differs then not too far from the elliptical distribution. In the strokes, still, the size and the change of the convinced drag place only a minor part. At least this shall be the case if the angle of prevalence at the sect root during the planting flight is kept constant. Advisably, the distribution of sect passion will be approached to the left distribution shape of the downcast stroke stop from the sect root to about the center of the sect half gauge the sect depth is nearly constant. From there to the sect tip the figure is phased.( For illustration like having of a duck, chump or seagull).

**5. COMPONENTS USED**

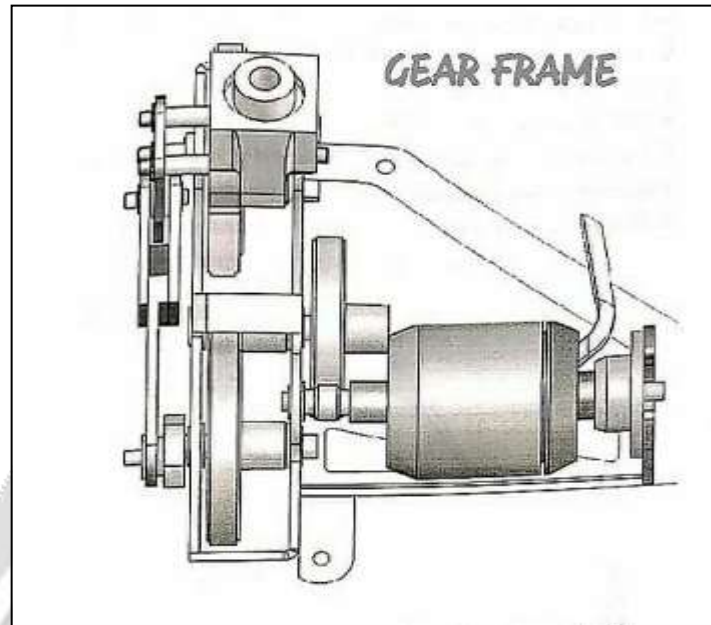
**5.1 Body used**



**Fig-5.1:** Mechanism body

The body is made from the carbon fibre frame where the electronic factors are attached. The places are cut out predicated on the dimension of each factors and fated locales. These Effective places are formerly cut help keep the electronic factors fixed in place. The carbon frame is used as it's feathery and has high durability in the presence of slightly strong wind.

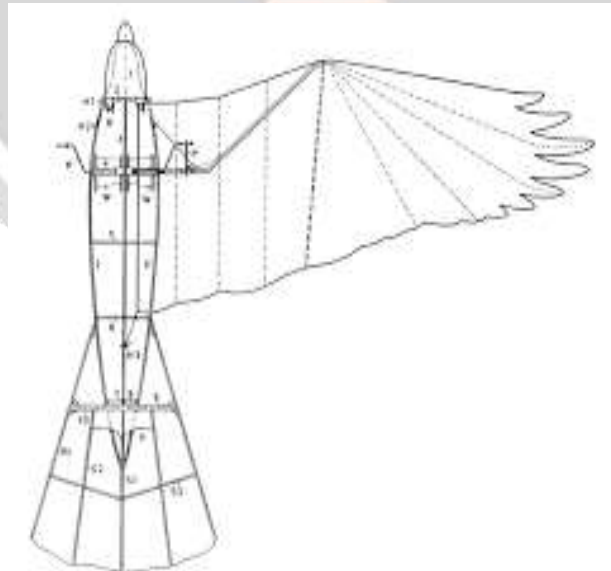
## 5.2. Gear Mechanism



**Fig -5.2:** Gear mechanism

Its having a three gear and the brushless motor to .The gear ratio is depends on the speed and the lift that we required .So we want to design in a such a manner to lift the body with the weight so gear ratio is important for us .

## 5.3 Wings

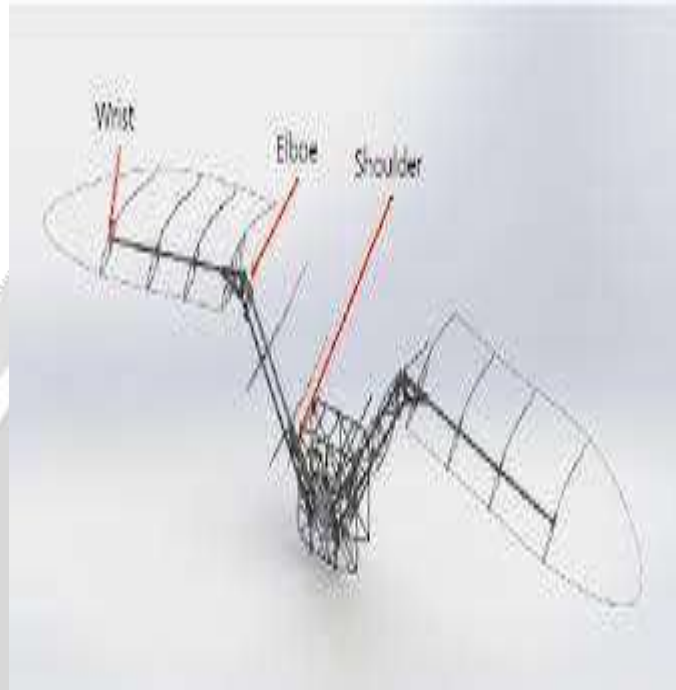


**Fig-5.3** wings design

In order to achieve the asked inflexibility and minimal weight, masterminds and experimenters have experimented with bodies that bear carbon fiber, plywood, fabric, and caricatures, with a stiff, strong running edge.( 38) Any mass

located posterior of the empennage reduces the sect's performance, so featherlight accoutrements and empty space are used where possible. To minimize drag and maintain the asked shape, choice of a material for the sect face is also important. In DeLaurier's trials, a smooth aerodynamic face with a double- face airfoil is more effective at producing lift than a single- face airfoil.. The Wingspan 27 elevation Length 20 elevation Gross Weight 55 grams for our model specification

#### 5.4 Overall view



**Fig-5.4:** overall view of wings

The single conrod medium employed in the Behemoth models. The sect spars form the commanding edge of the bodies and hold the bodies.



## 6. CALCULATIONS

The calculations are predicated on the data acquired from the factors used,

Dc motor – 3700 kV or 4200 kV

Battery –7.4 V or 11.1 V

1) The speed of the model can be determined by the data attained, Speed = 3700 × 7.4

= 444 Rps.

Speed = 4200 × 11.1

= 777 Rps.

2) Gear teeth rate of the gear medium is demanded to find out the flopping rate for the motor, W.K.T  $n_1 = 444$  Rps.  
 $n_2 = ?$

Using the formula, (1)

Using the gear teeth of 72, 8, 9 and 84.  $444/n_2 = 84/9$   $n_2 = 47.57$  Rps.

therefore,

=  $n_2/n_3 = 72/8$   $n_3 = 5.285$  Rps.

Hence, the number of flaps are set up to be 5.285 flaps per second.

3) For the battery used in mAh, it depends on the model, If the average current drawn is 15 A for 10 beats, also

$c = I \times t$ . (2)

$15 \times 10 / 60 = 2500$  mAh.

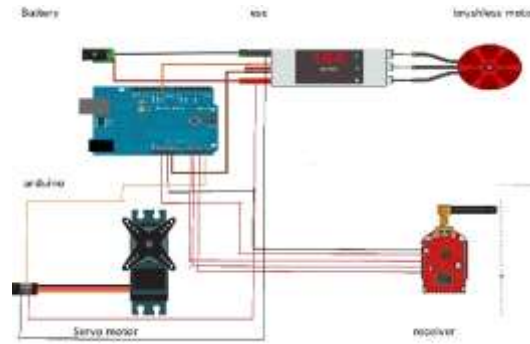
4) For current rate 'C', If peak current drawn is 30 A from 3000 mAh

$C = 30 \text{ A} / 3000 \text{ mAh}$

= 10C.

## 7. ELECTRONIC CIRCUIT

Electronic circuit done for these ornithopter is substantially on the transmitter and receiver as the prototype. substantially its has correspond the arduino, servo motor, electronic speed regulator, brushless motor are each connected to the receiver and for the power they all connected to the Li battery



**Fig-7.1:** Circuit connections on the ornithopter.



**Fig-7.2:** the remote controller.

The operation of prototype will work on the basics of simple RF signal. well transmitter will shoot the signal and The receiver will admit the signal and operates the the ornithopter will help in the flaps the bodies according to the given speed

## 8. CONCLUSION

Ornithopters, also known as flapping-wing aircraft, are a fascinating type of flying machine that mimic the flight of birds by flapping their wings. While there have been various attempts to build functional ornithopters throughout history, it remains a challenging feat due to the complex aerodynamics and mechanics involved. Despite the difficulties, recent advancements in technology and materials have led to significant progress in ornithopter design and construction. Ornithopters have potential applications in fields such as surveillance, search and rescue, and environmental monitoring. Additionally, they have the potential to be more energy-efficient than traditional fixed-wing aircraft, making them an attractive option for certain applications. However, there are still significant challenges that need to be addressed, including power supply, control, stability, and noise reduction. Overall, while ornithopters are an exciting area of research, there is still much work to be done before they can be used reliably and widely in practical applications





**Fig-8.1:** Prototype.

The following conclusions are drawn from the work

1. Surveillance is carried out using the model, images are transferred back taken by the lens attachment to the system.
2. Use of RF signal helps in the wide range operations.
3. Sonar sensors range the distance between the model and forthcoming obstacles.
4. The model is able of generating low frequency of ultra-sonic swells to scarify down catcalls from fields and field runways. This generated frequency isn't dangerous to catcalls.

## 9. ACKNOWLEDGEMENT

I would also like to acknowledge the contributions of the engineers, designers, and researchers who have continued to refine and improve ornithopter technology over the years, making it more efficient, maneuverable, and versatile. Lastly, I am grateful for the opportunity to learn about ornithopters and their history, and for the support and guidance of my mentors and colleagues who have helped me in my research on this fascinating topic.

## 10. REFERENCES

- [1] Zachary John Jackowski, "Design and Construction of an Autonomous Ornithopter" mainly based on Micro Aerial Vehicles (MAV) was published on 2009.
- [2] John H Lienhard V, "Contribution of Aerodynamic performance like thrust, lift and the range of magnitude of respective forces" published on 2011.
- [3] Stanley Seunghoon Baek, "Autonomous Ornithopter Flight with Sensor-Based Seeking Behaviour" published on 2011.
- [4] John wood, Daniel Jensen, Peter Leetsman, Christopher Gurrola, Daniel Zheng, Mathew Sparta and Richard Culver, Timothy Philpot, "Development of next generation Ornithopter prototypes like both virtual and physical 3axis rotational wing motion" published by on 2016.
- [5] Carlos Berdeguer, Hanna Schmidtman, "Effect on length of wings and limited forces on wings and linkage capable of rotating in all directions and translating the motion of four-bar linkage" published on 2015.